

Guided Tour

Chapter Introduction. Each chapter begins with an introductory section that sets up the purpose and goals of the chapter, describing in simple terms the material that will be covered and its application to the solution of engineering problems. Chapter Objectives provide students with a preview of chapter topics.

Chapter Lessons. The body of the text is divided into units, each consisting of one or several theory sections, Concept Applications, one or several Sample Problems, and a large number of homework problems. The Companion Website contains a Course Organization Guide with suggestions on each chapter lesson.

Concept Applications. Concept Applications are used extensively within individual theory sections to focus on specific topics, and they are designed to illustrate specific material being presented and facilitate its understanding.

Sample Problems. The Sample Problems are intended to show more comprehensive applications of the theory to the solution of engineering problems, and they employ the SMART problem-solving methodology that students are encouraged to use in the solution of their assigned problems. Since the sample problems have been set up in much the same form that students will use in solving the assigned problems, they serve the double purpose of amplifying the text and demonstrating the type of neat and orderly work that students should cultivate in their own solutions. In addition, in-problem references and captions have been added to the sample problem figures for contextual linkage to the step-by-step solution.

Homework Problem Sets. Over 25% of the nearly 1500 homework problems are new or updated. Most of the problems are of a practical nature and should appeal to engineering students. They are primarily designed, however, to illustrate the material presented in the text and to help students understand the principles used in mechanics of materials. The problems are grouped according to the portions of material they illustrate and are arranged in order of increasing difficulty. Answers to a majority of the problems are given at the end of the book. Problems for which the answers are given are set in blue type in the text, while problems for which no answer is given are set in red.



Concept Application 1.1

Considering the structure of Fig. 1.1 on page 5, assume that rod *BC* is made of a steel with a maximum allowable stress $\sigma_{all} = 165 \text{ MPa}$. Can rod *BC* safely support the load to which it will be subjected? The magnitude of the force F_{BC} in the rod was 50 kN. Recalling that the diameter of the rod is 20 mm, use Eq. (1.5) to determine the stress created in the rod by the given loading.

$$P = F_{BC} = +50 \text{ kN} = +50 \times 10^3 \text{ N}$$

$$A = \pi r^2 = \pi \left(\frac{20 \text{ mm}}{2} \right)^2 = \pi (10 \times 10^{-3} \text{ m})^2 = 314 \times 10^{-6} \text{ m}^2$$

$$\sigma = \frac{P}{A} = \frac{+50 \times 10^3 \text{ N}}{314 \times 10^{-6} \text{ m}^2} = +159 \times 10^6 \text{ Pa} = +159 \text{ MPa}$$

Since σ is smaller than σ_{all} of the allowable stress in the steel used, rod *BC* can safely support the load.

Sample Problem 1.2

The steel tie bar shown is to be designed to carry a tension force of magnitude $P = 120 \text{ kN}$ when bolted between double brackets at *A* and *B*. The bar will be fabricated from 20-mm-thick plate stock. For the grade of steel to be used, the maximum allowable stresses are $\sigma = 175 \text{ MPa}$, $\tau = 100 \text{ MPa}$, and $\sigma_b = 350 \text{ MPa}$. Design the tie bar by determining the required values of (a) the diameter *d* of the bolt, (b) the dimension *b* at each end of the bar, and (c) the dimension *h* of the bar.

STRATEGY: Use free-body diagrams to determine the forces needed to obtain the stresses in terms of the design tension force. Setting these stresses equal to the allowable stresses provides for the determination of the required dimensions.

MODELING and ANALYSIS:

a. Diameter of the Bolt. Since the bolt is in double shear (Fig. 1), $F_1 = \frac{1}{2}P = 60 \text{ kN}$.

$$\tau = \frac{F_1}{A} = \frac{60 \text{ kN}}{\frac{1}{2}\pi d^2} = 100 \text{ MPa} = \frac{60 \text{ kN}}{\frac{1}{2}\pi d^2} \quad d = 27.6 \text{ mm}$$

Use $d = 28 \text{ mm}$

At this point, check the bearing stress between the 20-mm-thick plate (Fig. 2) and the 28-mm-diameter bolt.

$$\sigma_b = \frac{P}{td} = \frac{120 \text{ kN}}{(0.020 \text{ m})(0.028 \text{ m})} = 214 \text{ MPa} < 350 \text{ MPa} \quad \text{OK}$$

b. Dimension *b* at Each End of the Bar. We consider one of the end portions of the bar in Fig. 3. Recalling that the thickness of the steel plate is $t = 20 \text{ mm}$ and that the average tensile stress must not exceed 175 MPa, write

$$\sigma = \frac{\frac{1}{2}P}{td} = 175 \text{ MPa} = \frac{60 \text{ kN}}{(0.020 \text{ m})d} \quad a = 17.14 \text{ mm}$$

$$b = d + 2a = 28 \text{ mm} + 2(17.14 \text{ mm}) \quad b = 62.3 \text{ mm}$$

c. Dimension *h* of the Bar. We consider a section in the central portion of the bar (Fig. 4). Recalling that the thickness of the steel plate is $t = 20 \text{ mm}$, we have

$$\sigma = \frac{P}{th} = 175 \text{ MPa} = \frac{120 \text{ kN}}{(0.020 \text{ m})h} \quad h = 34.3 \text{ mm}$$

Use $h = 35 \text{ mm}$

REFLECT and THINK: We sized *d* based on bolt shear, and then checked bearing on the tie bar. Had the maximum allowable bearing stress been exceeded, we would have had to recalculate *d* based on the bearing criterion.

Chapter Review and Summary. Each chapter ends with a review and summary of the material covered in that chapter. Subtitles are used to help students organize their review work, and cross-references have been included to help them find the portions of material requiring their special attention.

Review Problems. A set of review problems is included at the end of each chapter. These problems provide students further opportunity to apply the most important concepts introduced in the chapter.

Review Problems

1.59 In the marine crane shown, link CD is known to have a uniform cross section of 50×150 mm. For the loading shown, determine the normal stress in the central portion of that link.

Fig. P1.59

1.60 Two horizontal 5-kip forces are applied to pin B of the assembly shown. Knowing that a pin of 0.8-in. diameter is used at each connection, determine the maximum value of the average normal stress (a) in link AB, (b) in link BC.

Fig. P1.60

1.61 For the assembly and loading of Prob. 1.60, determine (a) the average shearing stress in the pin at C, (b) the average bearing stress at C in member BC, (c) the average bearing stress at B in member BC.

Computer Problems. Computers make it possible for engineering students to solve a great number of challenging problems. A group of six or more problems designed to be solved with a computer can be found at the end of each chapter. These problems can be solved using any computer language that provides a basis for analytical calculations. Developing the algorithm required to solve a given problem will benefit the students in two different ways: (1) it will help them gain a better understanding of the mechanics principles involved; (2) it will provide them with an opportunity to apply the skills acquired in their computer programming course to the solution of a meaningful engineering problem.

Review and Summary

This chapter was devoted to the concept of stress and to an introduction to the methods used for the analysis and design of machines and load-bearing structures. Emphasis was placed on the use of a *free-body diagram* to obtain equilibrium equations that were solved for unknown reactions. Free-body diagrams were also used to find the internal forces in the various members of a structure.

Axial Loading: Normal Stress
The concept of *stress* was first introduced by considering a two-force member under an axial loading. The normal stress in that member (Fig. 1.41) was obtained by

$$\sigma = \frac{P}{A} \quad (1.5)$$

The value of σ obtained from Eq. (1.5) represents the *average stress* over the section rather than the stress at a specific point Q of the section. Considering a small area ΔA surrounding Q and the magnitude ΔF of the force exerted on ΔA , the stress at point Q is

$$\sigma = \lim_{\Delta A \rightarrow 0} \frac{\Delta F}{\Delta A} \quad (1.6)$$

In general, the stress σ at point Q in Eq. (1.6) is different from the value of the average stress given by Eq. (1.5) and is found to vary across the section. However, this variation is small in any section away from the points of application of the loads. Therefore, the distribution of the normal stresses in an axially loaded member is assumed to be *uniform*, except in the immediate vicinity of the points of application of the loads.

For the distribution of stresses to be uniform in a given section, the line of action of the loads P and P' must pass through the centroid C . Such a loading is called a *centric axial loading*. In the case of an *eccentric axial loading*, the distribution of stresses is *not uniform*.

Transverse Forces and Shearing Stress
When equal and opposite *transverse forces* P and P' of magnitude P are applied to a member AB (Fig. 1.42), *shearing stresses* τ are created over any section located between the points of application of the two forces.

Fig. 1.41 Axially loaded member with cross section normal to member used to define normal stress.

Fig. 1.42 Model of transverse resultant forces on either side of C resulting in shearing stress at section C.

Computer Problems

The following problems are designed to be solved with a computer.

1.C1 A solid steel rod consisting of n cylindrical elements welded together is subjected to the loading shown. The diameter of element i is denoted by d_i and the load applied to its lower end by P_i , with the magnitude P_i of this load being assumed positive if P_i is directed downward as shown and negative otherwise. (a) Write a computer program that can be used with either SI or U.S. customary units to determine the average stress in each element of the rod. (b) Use this program to solve Probs. 1.1 and 1.3.

1.C2 A 20-kN load is applied as shown to the horizontal member ABC. Member ABC has a 10×50 -mm uniform rectangular cross section and is supported by four vertical links, each of 8×36 -mm uniform rectangular cross section. Each of the four pins at A, B, C, and D has the same diameter d and is in double shear. (a) Write a computer program to calculate for values of d from 10 to 30 mm, using 1-mm increments, (i) the maximum value of the average normal stress in the links connecting pins B and D, (ii) the average normal stress in the links connecting pins C and E, (iii) the average shearing stress in pin B, (iv) the average shearing stress in pin C, (v) the average bearing stress at B in member ABC, and (vi) the average bearing stress at C in member ABC. (b) Check your program by comparing the values obtained for $d = 16$ mm with the answers given for Probs. 1.7 and 1.27. (c) Use this program to find the permissible values of the diameter d of the pins, knowing that the allowable values of the normal, shearing, and bearing stresses for the steel used are, respectively, 150 MPa, 90 MPa, and 230 MPa. (d) Solve part c, assuming that the thickness of member ABC has been reduced from 10 to 8 mm.

Fig. P1.C1

Fig. P1.C2

List of Symbols

a	Constant; distance	P_U	Ultimate load (LRFD)
A, B, C, ...	Forces; reactions	q	Shearing force per unit length; shear flow
A, B, C, \dots	Points	Q	Force
A, \bar{a}	Area	Q	First moment of area
b	Distance; width	r	Radius; radius of gyration
c	Constant; distance; radius	R	Force; reaction
C	Centroid	R	Radius; modulus of rupture
C_1, C_2, \dots	Constants of integration	s	Length
C_p	Column stability factor	S	Elastic section modulus
d	Distance; diameter; depth	t	Thickness; distance; tangential deviation
D	Diameter	T	Torque
e	Distance; eccentricity; dilatation	T	Temperature
E	Modulus of elasticity	u, v	Rectangular coordinates
f	Frequency; function	u	Strain-energy density
F	Force	U	Strain energy; work
$F.S.$	Factor of safety	v	Velocity
G	Modulus of rigidity; shear modulus	V	Shearing force
h	Distance; height	V	Volume; shear
H	Force	w	Width; distance; load per unit length
H, J, K	Points	W, W	Weight, load
I, I_x, \dots	Moment of inertia	x, y, z	Rectangular coordinates; distance; displacements; deflections
I_{xy}, \dots	Product of inertia	$\bar{x}, \bar{y}, \bar{z}$	Coordinates of centroid
J	Polar moment of inertia	Z	Plastic section modulus
k	Spring constant; shape factor; bulk modulus; constant	α, β, γ	Angles
K	Stress concentration factor; torsional spring constant	α	Coefficient of thermal expansion; influence coefficient
l	Length; span	γ	Shearing strain; specific weight
L	Length; span	γ_D	Load factor, dead load (LRFD)
L_e	Effective length	γ_L	Load factor, live load (LRFD)
m	Mass	δ	Deformation; displacement
M	Couple	ϵ	Normal strain
M, M_x, \dots	Bending moment	θ	Angle; slope
M_D	Bending moment, dead load (LRFD)	λ	Direction cosine
M_L	Bending moment, live load (LRFD)	ν	Poisson's ratio
M_U	Bending moment, ultimate load (LRFD)	ρ	Radius of curvature; distance; density
n	Number; ratio of moduli of elasticity; normal direction	σ	Normal stress
p	Pressure	τ	Shearing stress
P	Force; concentrated load	ϕ	Angle; angle of twist; resistance factor
P_D	Dead load (LRFD)	ω	Angular velocity
P_L	Live load (LRFD)		